

## FEDERAL AVIATION AGENCY

E-143  
Revision 4  
Pratt & Whitney  
WASP S3H1  
August 7, 1973

## AIRCRAFT ENGINE SPECIFICATION

Engines of models described herein conforming with this specification and approved data on file with the Federal Aviation Agency meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft specifications and applicable portions of the Civil Air Regulations provided they are installed, operated, and maintained as prescribed by the approved manufacturer's manuals and other approved instructions.

Manufacturer Pratt & Whitney Aircraft  
Division of United Aircraft Corporation  
East Hartford, Connecticut

Model	Wasp	S3H1
Type	9RA - Direct drive	
Rating (Impeller gear ratio)	10:1	
Maximum continuous, hp., rpm, in.Hg., at:	550-2200-32.5-5000	
Critical altitude (ft.)	550-2200-34.0-S.L.	
Sea level pressure altitude (ft.)	600-2250-35.5-3000	
Takeoff (5 minutes), hp., rpm, in.Hg., at:	600-2250-36.0-S.L. (See Note 5)	
Critical altitude (ft.)		
Sea level pressure altitude (ft.)		
Fuel (minimum grade aviation gasoline)	80/87	
Lubricating oil	See P&WA Service Bulletin No. 1183	
Bore and stroke, in.	5.75 X 5.75	
Displacement, cu. in.	1344	
Compression ratio	6:1	
Weight (dry), lb.	865	
Propeller shaft, SAE No.	40	
Carburetion	Stromberg NA-Y9B, C, H, E1 with 2-3/4 in. venturis or PD-9D1 with 2-1/2 in. venturis	
Ignition, Dual	Scintilla SB9R or Bosch SB9RU2 and SB9RU3 magnetos	
Timing, °BTC	25	
Spark plugs	See Note 6	
Notes	1, 2, 3, 4, 5, 6	

Certification basis Type Certificate No. 143

Production basis Production Certificate No. 2

NOTE 1. Maximum permissible temperatures are as follows:

<u>Cylinder Head</u>	<u>Cylinder Barrel</u>	<u>Oil Inlet</u>
500°F.	300°F.	200°F.

NOTE 2. Fuel and oil pressure limits:

Fuel pressure (psi)	3-6
Oil pressure (psi)	70-90

NOTE 3. The following accessory drives are provided:

	Direction of Rotation (Clockwise or Counter-Clockwise)	Speed Ratio (Times Crankshaft Speed)	Maximum Torque (in. lb.)		**Maximum Overhang (in.-lb.)
			Cont.	Static	
Starter	CC	1.0	2300	10,000	180
Generator	C	1.5	100	300	183
Generator (opt.)	C	2.0	150	900	183
Fuel pump	CC	1.0	150	450	10
Aux. accessory	CC	1.0	*150	450	10
Tachometer R.H.	C	.5	15	45	10
L.H.	CC	.5	15	45	10
Governor	C	1.144	60	250	-
Vacuum pump	C	1.5	*100	300	10

\*Maximum combined drive load of vacuum pump and fuel pump must not exceed 150 in.-lb.

\*\*Maximum allowable accessory moments in in.-lb., provided no destructive accessory drive or mounting pad forces resulting from accessory vibration are present.

Early Wasp engines incorporate tongue and groove vacuum and fuel pump drives. All later engines have spline type drives.

NOTE 4. Eligible with automatic power and mixture control when the NA-Y9C carburetor is used at a weight increase of 12 pounds.

NOTE 5. (a) Ratings are based upon the best power mixture strength and 450°F cylinder head temperature.

(b) With the use of grade 91/96 fuel, the following ratings may be used:

Maximum continuous, hp, rpm, in.Hg., at:

Critical altitude (ft.)

Sea level pressure altitude (ft.)

Takeoff (one minute), hp, rpm, in.Hg., at:

Critical altitude (ft.)

Sea level pressure altitude (ft.)

550-2200-33.0-4500

550-2200-34.0-S.L.

600-2250-36.0-2300

600-2250-36.5-S.L.

No increase in power by using higher octane fuel

NOTE 6. The following spark plugs are approved on this engine:

AC SR-83P, S-86R, HSR-86, HSR-83P

Autolite SH-2K, SH-2M, SH-20

BG 321S, 417S, SS485A, RB485S, 706SR

Champion C27S, ED41N, EM41N, RC26S, RED39N, REM39N, RHD39M, RHM39N

.....END.....